N613ES 2005 Cessna T182T

FAA Form 337s

MSN: T18208424



Prepared by the worldwide aviation specialists at RidgeAire, Inc.

US Department
of Transportation
Federal Aviation
Administration

MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

OMB No. 2120-0020 Exp: 5/31/2018	Electronic Tracking Number			
	or FAA Use Only			

Feder	ral Aviation raistration	on	(All	IIIali	I U , P	owerplant, Pro	·Þα	ener, or a	whh	ianc e)						
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2. Ow	ner	LANS	FOF	RD C) IR	ESON JR	₹		City	Houston	Jess Di. 31E	230	·	State TX	_	
										Zip	77070			Country	US	_
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Rep	pair	Alteration		Unit			lak	к е				Model			Serial No.	
		x	AIRE	RAM	E	CESSNA			(As described in Item 1 above)			е)	T18208424			
	_		POV	VERP	LANT											
			PRC	PELL	ER											
			APP	LIAN	Œ	Type Manufacturer										
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A. Age		Name and A	ddres	s			4	B. Kind of		ated Mecha	nic		1 1	Manuf	acturer	_
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City	Sugar			us		State TX	Ţ			Repair Statio			OSJF	R598L Radio	o Class I, II, III , Limited Inst, Limited Airfran	ne
ı	have be	that the rep	air an	d/or a	with rect to	the requirements of the best of my known	ř P owl	identified i Part 43 of the ledge.	n item e U.S		and des	cribed on t			or attachments hereto the information	_
	ded ran CFR F				Signa	ature/Date of Author	U.	er		9-1-		2018	_			
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						ns specified below ministration and is	/ ,	tne unit i		ied in iter X Approve		Rejecte		in the	manner prescribed by the	;
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Certific Design	ate or ation N	o. OSJR	598	L	Signa	nture/Date of Author	rize	ed Individua		a	1 2	C				-
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FAA Form 337 (10/06)

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Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished (If more space is required, attach additional sheets. Identify with aircraft national sheets.	onality and registration mark and da CESSNA T182T	te work completed.) T18208424
	N613ES	09/13/2018
	Nationality and Registration Mark	Date
Installed Garmin GTX 345R mode S transponder and	Garmin GA 35 WAAS (GPS Antenna.
Removed: GTX 33 w/rack 011-00779-10 89103968		
Installed: GTX 345R w/rack P/N 011-03303-40 S/N 3EH4204 GA 35 P/N 013-00235-00 S/N 155156	34	
This GTX 345R is installed in accordance with STC N system was shown to meet the equipment performant STC includes FAA Approved Airplane Flight Manual S 190-00734-15, Rev. 3 dated 12/21/2017 in the Pilot's Reference Instructions for Continued Airworthiness, G Rev. 6 dated 12/21/2017.	ce requirements of 14 C Supplement, Garmin Do Operating Handbook.	FR part 91.227. The ocument No.
The GA 35 is installed in accordance with STC No. SA Approved Antenna Installation Instructions, Garmin Do. 07/25/2014. Reference Instructions for Continued Airw 005-00638-09 Revision 3 dated 03/16/2012.	ocument No. 190-00848	3-00, Revision F dated
The aircraft Weight and Balance and Equipment list h	ave been amended.	
All systems have been satisfactorily tested for electric	al or magnetic interfere	nce.
Electrical load analysis of the aircraft has been perform	med.	
This work has been accomplished under Sugar Land	Avionics, LP Work Orde	er 2452-09-2018
**************************************	*******	*******
Additional Sheets	Are Attached	

Department of Transportation -- Federal Aviation Administration

Supplemental Type Certificate

Number SA02018SE-D

This certificate issued to

Garmin International, Inc. 1200 East 151st Street Olathe, KS 66062

certifies that the change in the type design for the following product with the limitations and conditions therefore as specified hereon meets the airworthiness requirements of Part 23* of the Federal Aviation Regulations.

Criginal Freduct Type Certificate Number: *See attached Approved Model List (AML) NO. SA02018SE-D dated November 28, 2011 or later FAA approved revision for list of approved aircraft models and applicable airworthiness regulations.

Description of Type Design Change: Provisions installation of Garmin GPS/XM Antennas in accordance with FAA Approved Model List (AML) SA02018SE-D, dated November 28, 2011, or later FAA approved revision, and FAA Approved GPS/XM Antenna STC Master Drawing List, Document No.: 005-00638-01, Revision "1", approved on January 4, 2011, or later FAA approved revision.

Bimitutions and Conditions:

- Compatibility of this design change with previously approved modifications must be determined by the installer.
- 2. A copy of this certificate must be maintained as part of the permanent records for the modified aircraft.
- 3. If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission.
- 4. Aircraft modified by this STC must be maintained in accordance with the Instructions for Continued Airworthiness (ICA) listed in the MDL identified above.

This certificate and the supporting data which is the basis for approval shall remain in offect until surrendered, suspended, revoked or a termination date is otherwise established by the Administrator of the Tederal Aviation Administration.

Late of application: July 22, 2010

2010 Sute reissued: November 28, 2011

Sale of issuance: January 4, 2011

Lute umended:



By direction of the Administrator

(Signature)

Robert M. Grove
ODA STC Unit Administrator
ODA-240087-CE

Garmin International, Inc.

(Title)

INSTRUCTIONS: The transfer endorsement below may be used to notify the appropriate FAA Regional Office of the transfer of this Supplemental Type Certificate. The FAA will reissue the certificate in the name of the transferee and forward it to him. Transfer the ownership of Supplement Type Certificate Number To I. Name of transferred (. Number and street) (City. State and Zip code) From & Sume of grunters Print or type (Address of granter) (Number and street) (City. State and Zip code) Extent of Authority (if licensing agreement): Date of Transfer: Signature of grantor /Ju Juli:

• Any alteration of this pertificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years, or outs.

This certificate may be transferred in accordance with FAR 31.47.

FAA FGRI 8110-2(10-68) PAGE 2 of 3 PAGES

Department of Transportation - Federal Abiation Administration

Supplemental Type Certificate

(Continuation Sheet)

Number SA02018SE-D

Date of Issuance: November 28, 2011

*Certification Basis

Based on 14 CFR §§ 21.115 and 21.101, and the FAA policy for significant changes in FAA Order 8110.48, the certification basis for this change is as follows:

- a. The certification basis for parts not changed or not affected by this change is shown on the attached AML.
- b. The certification basis for parts changed or affected by this change is:

Regulations at the latest amendment: No Amdt through 23-59

Section	Amdt	Section	Amdt
23.251	23-45	23.609	No Amdt
23.301(a)(b)(c)	23-48	23.611	23-48
23.303	No Amdt	23.613(a)	23-45
23.305(a)	23-45	23.867	23-49
23.307(a)	No Amdt	23.1301	23-20
23.571(c)*	23-48	23.1308(a)(b)(c)	23-57
23.573(b)*	23-48	23.1309(a)(1)(b)(1)(e)	23-49
23.603	23-23	23.1431(a)(b)	23-49
23.605(a)	23-23	23.1529	23-26
23.607(a)	23-48		



March 2, 2011

Subject:

STC Permission to use STC SA02018SE-D for Garmin Model GPS/XM Antennas AML STC

(See AML)

Consistent with Order 8110.4B and AC21-40, Garmin International grants permission to Garmin dealers, installers, and owners of the Garmin Model GPS/XM Antennas AML STC units to use STC SA02018SE-D and deliverable data associated with it, for the sole and express purpose of installation and approval of the installation of the Garmin Model GPS/XM Antennas AML STC, and associated interfaces to other previously approved equipment.

Emmett Griffith, Engineering Manager

Garmin AT, Inc.

1200 East 151st Street Olathe, KS 66062 P: 913-397-8200 F: 913-397-8282

March 8, 2016

Subject:

STC Permission to use STC SA01714WI for

Garmin GTX 330, 330D, 33, 33D, 335, and 335R with ADS-B Out (see AML);

Garmin GTX 345 and 345R with ADS-B In and Out (see AML)

Consistent with FAA Order 8110.4() and Advisory Circular (AC) 21-40(), Garmin International grants permission to Garmin dealers, installers, and owners of the Garmin GTX 330, 330D, 33, 33D, 335, 335R, 345, and 345R to use STC SA01714WI and deliverable data associated with it, for the sole and express purpose of installation and approval of the installation of the Garmin GTX 330, 330D, 33, 33D, 335, 335R, 345, and 345R and associated interfaces to other previously approved equipment.

Emmett Griffith

Director of Engineering

Garmin AT, Inc.

Department of Transportation -- Federal Aviation Administration

Supplemental Type Certificate

Number SA01714WI

This certificate issued to

Garmin International, Inc. 1200 East 151st Street Olathe, KS 66062

certifies that the change in the type design for the following product with the limitations and conditions therefore as specified hereon meets the airworthiness requirements of Part 23* of the Federal Aviation Regulations.

Original Product - Type Certificate

Number: Moake: * See attached Approved Model List (AML) No. SA01714WI dated May 1, 2013 or later FAA-approved revision for list of approved aircraft models and applicable airworthiness regulations.

Description of Type Design Change:

Installation of a Garmin transponders: (a) GTX 330/330D/33/33D with ADS-B Out functionality; (b) GTX 335/335R/345/R with ADS-B Out and In functionality; or (c) GTX 335R/345R with ADS-B Out functionality in select airplanes installed with G950/G1000 systems.

Data Required:

- (1) Garmin Master Drawing List (MDL) 005-00734-04, Revision 1, dated May 1, 2013 or later FAA-approved revision.
- (2) Garmin Airplane Flight Manual Supplement or Supplemental Airplane Flight Manual (AFMS), 190-00734-15, Revision 1, dated May 1, 2013 or later FAA-approved revision.

Limitations and Conditions:

- (1) Compatibility of this design change with previously approved modifications must be determined by the installer.
- (2) Aircraft installations involving the Garmin transponder GTX 33/33D/330/330D models require the previous installation of an approved ADS-B position source. Refer to the design data specified in the Master Drawing List (MDL) listed above for specific hardware and software requirements.
- (3) Aircraft installations involving the Garmin transponder GTX 335/335R/345/345R models without an optional internal GPS require the previous installation of an approved ADS-B position source. Refer to the design data specified in the Master Drawing List (MDL) listed above for specific hardware and software requirements.

This certificate and the supporting data which is the basis for approval shall remain in offect until surrendered, suspended, revoked or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of application: August 7, 2012

Date reissued:

Dute of issuance: May 1, 2013

Date amended: April 29, 2014, March 8, 2016

By direction of the Administrator

TOMINISTRATION

Mahay Whe

Michael Warren
ODA STC Unit Administrator
ODA-240087-CE
Garmin International, Inc.

(Title)

Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years, or both.

INSTRUCTIONS: The transfer endorsement below may be used to notify the appropriate FAA Regional Office of the transfer of this Supplemental Type Certificate. The FAA will reissue the certificate in the name of the transferee and forward it to him. Transfer the ownership of Supplement Type Certificate Number To (Name of transferce) (City, State and Zip code) From (Name of granter) Print or type (Number and street) (Address of granter) (City, State and Zip code) Extent of Authority (if licensing agreement): Date of Transfer: Signature of grantor /In Ink/.

Department of Transportation -- Federal Aviation Administration

Supplemental Type Certificate

(Continuation Sheet)

Number SA01714WI

Date of issuance: May 1, 2013

Date Amended: April 29, 2014, March 8, 2016

Limitations and Conditions (continued):

- (4) Aircraft with G950 or G1000 display systems are limited to Garmin transponder GTX 335R/345R models; see Table II of the attached AML. Refer to the design data specified in the Master Drawing List (MDL) listed above for specific hardware and software requirements.
- (5) Aircraft modified by this STC must be operated in accordance with the Airplane Flight Manual Supplement or Supplemental Airplane Flight Manual (AFMS) identified above.
- (6) Aircraft modified by this STC must be maintained in accordance with the Instructions for Continued Airworthiness (ICA) listed in the Master Drawing List (MDL) identified above.
- (7) If the holder agrees to permit another person to use this certificate to alter the product, the holder must give the other person written evidence of that permission.

*Certification Basis

Based on 14 CFR §§ 21.115 and 21.101, and the FAA policy for significant changes in FAA Order 8110.48, the certification basis for this change is as follows:

- a. The certification basis for parts not changed or not affected by this change is shown on the attached AML.
- b. The certification basis for parts changed or affected by this change is:

Section	Amdt	Section	<u>Amdt</u>	Section	Amdt
23.301 a	23-48	23.773 a(2)	23-45	23.1357 a, b, c, d	23-43
23.303	Original	23.777 a, b	23-62	23.1359 с	23-49
23.305	23-45	23.853 a	23-62	23.1365 a, b, d, e	23-49
23.307 a	Original	23.863 a, b	23-34	23.1367 a, b, c, d	Original
23.337	23-48	23.1301 a, b, c	23-62	23.1431 a, b	23-62
23.561 b(3), e	23-62	23.1306 a, b	23-61	23.1501 a, b	23-21
23.601	Original	23.1307	23-49	23.1525	23-45
23.603	23-23	23.1308 a, b, c	23-57	23.1529	23-26
23.605 a	23-23	23.1309 a(1)(2),b, c, d	23-62	23.1555 a, b, e(2)	23-62
23.607 a, b	23-48	23.1310	23-62	23.1581 a, b, c, f	23-50
23.611	23-48	23.1311 a(2)	23-62	23.1583 h, m	23-62
23.613 a, b	23-45	23.1322 b, d, e	23-43	23.1585 ј	23-62
23.627	Original	23.1351 a	23-49		
23.771 a	23-14	23.1353 h	23-62		
		END			

(3)	
U.S. Department of Transportation	
Federal Aviation Administration	

Form Approved Electronic Tracking Number OMB No.2120-0020

(C)			MA	JOR REPAIR AN	D A	ALTERA'	TION		2/28/2011		FAA Haa Ool	
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Federal Aviation Administration INSTRUCTIONS: Print or type all entries. See Title CFR §43.9							· ·					
instruc	tions an	d disposi		rm. This report is require							ent revision thereof) for result in a civil penalty for	
1. Ai	rcraft		ity and Registra I613ES	tion Mark		Serial No. T18208424						
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			PROPELLER					,				
			APPLIANCE	Type								
	1			6. Co	nfo	rmity State	 ment					
A. Ag	ency's f	Name and	d Address		T	B. Kind of						
Name	Andre	w C. Vaugl	hn .		. x	U.S. Certifica	ted Mechanic				Manufacturer	
Address	834 M	ladene Dr.				Foreign Certi	ficated Mechai	nic		C. (Certificate No.	
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NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

	USA N613ES	02/18/2014
	Nationality and Registration Mark	Date
/2002. (Air Comm Corporation ort NO. CR-182-9 Airplane Flig	air conditioning system in accordance with Master Dr. STC # ST09494SC). Updated Pilots Operating Handb ht Manual Supplement Revision E dated: 8/30/2005. I For Continued Airworthiness are Contained in Docum	book with Keith Products L.P. Certification Updated Weight and Balance with Certification

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J.S.	Departmen

U.S. Department of Transportation Federal Aviation Administration

MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

	Electronic	Tracking	N	mber
OMB No.2120-0020	1		- 1	
11/30/2007				

For FAA Use Only

INSTRUCTIONS: Print or type all entries. See Title CFR 43.9, Part 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 44701). Failure to report can result in a civil penalty for each such violation (49 U.S.C. 46301(a)).

		49 U.S.C. 463		ins report is require	u by	149 U.	5.0. 44701). Fallur	е то героп	can re	esult in a civil penalty for
1. Aircraf	a. I	nality and Regis			Serial No. T18208424					
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	Make Cess					-	Model T182T		*	Series
	Name	e (As shown on r	egistratio	n certificate)	_		Address (As show	n on registrat	ion cert	tificate)
2. Owne	3 r	aft Guaranty (•	•			Address PO Box 2549			
							City Onalas			State TX
							Zip <u>77360-</u>	-2549 Cou	intry	United States of America
				3. Fo	r F	AA Use Onl	у			
Regulation	ns Part 21.	The Date idea	ntified he	red unless it has an a erein complies with th a person authroized i	ne ap	plicable airv	vorthiness requireme	ents and is a FA Da	ipprovi A Initi te <u>03</u>	f Federal Aviation ed for the above described al Approval -/3 -/3 0-15 TPA
4. Type 5. Uni					it Id	entification	1	Name		
Repair	Alteratio	n Unit		Make	,		Model			Serial No.
	x	AIRFRAME	AIRFRAME				(As described in item 1 above		θ)	
		POWERPL	.ANT	rige.						***************************************
		PROPELL	ER							
		APPLIANCE Type Manufacturer								
	•	•	•	6. Co	nfo	rmity State	ment			
A. Agenc	y's Name	and Address			T	B. Kind of	Agency			
Name Michael T. Martinetto						U.S. Certificated Mechanic				Manufacturer
Address 8	949 Fairwea	ther Dr.			Г	Foreign Certificated Mechanic			C. Certificate No.	
City <u>L</u>	argo		s	state FL		Certificated Repair Station			3259098	
Zip 33773 Country USA						Certificated Maintenance Organization				
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NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

Structural substantiation was provided in report WOE-02-08 dated 18 November 2002 and approved by 8110-3 dated 18 November 2002. The fuel system pass been operationally checked and the engine runs on all fuel tanks. The fuel system operates normally and loes not leak. ART ENCORACE The pilot has been furnished within "Airplane Auxiliary Fuel System Instructions" booklet for proper operations of the auxiliary fuel ystem for Cessna 182T. Upon arrival at the destination listed on the 8130-7 Ferry Permit, this Field Approved 337 expires and is void. The aircraft must be eturned to service and the auxiliary fuel system removed in accordance to the attached WOE 182RI removal instructions. When the aircraft has been returned to its original configuration, the aircraft has been inspected for damage while in the overweight ondition per Cessna 182 Service Manual, a certified mechanic must make a signed entry in the aircraft records attesting to the return	(If more space is required, attach addition	nal sheets. Identify with a	aircraft nationality and registration	mark and date work completed.
Installed temporary auxiliary long range fuel tank / tanks per Wings of Eagles drawings no WOE C50CP Sheets 1 & 2 Dated 20 Oct 2002 and WOE C 100 Sheets 1 & 2, 182-1 sheets 1-14 dated November 1,4, 2002, 182FL Sheets 1-4 Dated October 20, 2002. In occordance with FAA form 8110-3 dated 18 November 2002 provided by DETT- 51057-5CF, 4C, 43, 13-18 and accepted industry ractices pertaining to hardware and fittings. All tanks have been pressure checked to 3.5 PSI and found to be satisfactory for the tended installation. Introductal substantiation was provided in report WOE-02-08 dated 18 November 2002 and approved by 8110-3 dated 18 November 2002. The fuel system operations per advised in the engine runs on all fuel tanks. The fuel system operates normally and oes not leak. **System Provided Within **Air Plane Auxiliary Fuel System Instructions** booklet for proper operations of the auxiliary fuel system for Cessina** 182T. **Poon arrival at the destination listed on the 8130-7 Ferry Permit, this Field Approved 337 expires and is void. The aircraft must be alturned to service and the auxiliary fuel system removed in accordance to the altached WOE 182RI removal instructions. **When the aircraft has been returned to its original configuration, the aircraft has been inspected for damage while in the overweight on onlition per Cessina** 182 Ferry Earth and the aircraft records attesting to the return of service according to WOE 182RI. **NOTHING FOLLOWS**		N613ES		05/13/2011
9002 and WOE C 100 Sheets 1 & 2, 182-1 sheets 1-14 dated November 14, 2002, 182FL Sheets 1-1 Dated October 20, 2002. In coordance with FAA form 81 110-3 dated 18 November 2002 provided by DERT. 510375-CE, AC 43 13-18 and accepted industry ractices pertaining to hardware and fittings. All tanks have been pressure checked to 3.5 PSI and found to be satisfactory for the intended installation. Structural substantiation was provided in report WOE-02-08 dated 18 November 2002 and approved by 8110-3 dated 18 November 2002. The fuel system plans people of the provided by 8110-3 dated 18 November 2002. The fuel system page 300 per		Nationality and Registra	ation Mark	Date
002. The fuel system has been greationally checked and the engine runs on all fuel tanks. The fuel system operates normally and oes not leak. ACT FLOUR-COR he pilot has been furnished within "Airplane Auxiliary Fuel System Instructions" booklet for proper operations of the auxiliary fuel ystem for Cessña 182T. John arrival at the destination listed on the 8130-7 Ferry Permit, this Field Approved 337 expires and is void. The aircraft must be eturned to service and the auxiliary fuel system removed in accordance to the attached WOE 182RI removal instructions. When the aircraft has been returned to its original configuration, the aircraft has been inspected for damage while in the overweight ondition per Cessna 182 Service Manual, a certified mechanic must make a signed entry in the aircraft records attesting to the return or service according to WOE 182RI. ***********************************	2002 and WOE C 100 Sheets 1 & 2, 182-1 accordance with FAA form 8110-3 dated 18	sheets 1-14 dated Noven Novermber 2002 provide	nber 14, 2002, 182FL Sheets 1-4 E ed by DERT- 510375-CE. AC 43.1	Pated October 20, 2002. In 3-1B and accepted industry
Ipon arrival at the destination listed on the 8130-7 Ferry Permit, this Field Approved 337 expires and is void. The aircraft must be eturned to service and the auxiliary fuel system removed in accordance to the attached WOE 182RI removal instructions. When the aircraft has been returned to its original configuration, the aircraft has been inspected for damage while in the overweight ondition per Cessna 182 Service Manual, a certified mechanic must make a signed entry in the aircraft records attesting to the return or service according to WOE 182RI. ***********************************	002. The fuel system has been operationa oes not leak oten AपT	ally checked and the engi	ne runs on all fuel tanks. The fuel	system operates normally and
ondition per Cessna 182 Service Manual, a certified mechanic must make a signed entry in the aircraft records attesting to the return or service according to WOE 182RI. ***********************************	Jpon arrival at the destination listed on the	8130-7 Ferry Permit, this stem removed in accorda	Field Approved 337 expires and is not to the attached WOE 182RI re	s void. The aircraft must be moval instructions.
	When the aircraft has been returned to its o condition per Cessna 182 Service Manual, a conservice according to WOE 182RI.	riginal configuration, the a certified mechanic must	aircraft has been inspected for dar t make a signed entry in the aircraf	nage while in the overweight trecords attesting to the return
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